AD-A009 133

HYDROCARBON CONSTITUENTS OF T-56 COMBUSTOR EXHAUST

James P. Conkle, et al

School of Aeruspace Medicine Brooks Air Force Base, Texas

April 1975

DISTRIBUTED BY:



National Technical Information Service U. S. DEPARTMENT OF COMMERCE

This progress report was submitted by personnel of the Bioenvironmental Analysis Branch, Environmental Sciences Division, USAF School of Aerospace Medicine, AFSC, Brooks AFB, Texas under job order 7:64-16-13.

When U.S. Government drawings, specifications, or other data are used for any purpose other than a definitely related Government procurement operation. the Government thereby incurs no responsibility nor any obligation whatsoever: and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise, as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture. use, or sell any patented invention that may in any way be related thereto.

This report has been reviewed by the Information Office (OI) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nations.

This technical report has been reviewed and is approved for publication.

JAMES P. CONKLE, Ph. D.

Project Scientist

RICHARD L. MILLER, Ph. D. Project Scientist/Supervisor

EVAN R. GOLZAA. Colomel, UBAr. MC

Commander

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

REPORT DOCUMENTATION PAG	READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER	OVY ACCESSION NO. 3 RECIPIENT'S CATALOG NUMBER
SAM-TR-75-8	120-4009 133
4. TITLE (and Subtitle) HYDROCARBON CONSTITUENTS OF T-56	5 TYPE OF REPORT & PERIOD COVERED
COMBUSTOR EXHAUST	Progress Report Dec 1973 - Dec 1974
	6 PEFFORMING OPG. REPORT NUMBER
	THE PORT OR ME TO A TO
7. AUTHOR(s)	8 CONTRACT OR GRANT NUMBER(s)
James P. Conkle, Ph. D. William W. Lackey, B.S.	
Richard L. Miller, Ph. D.	
9. PERFORMING ORGANIZATION NAME AND ADDRESS	10. PROGRAM BUBMENT, PROJECT, TASK AREA & WORK UNIT TUMBERS
USAF School of Aerospace Medicine (VNL	AREA & WORK UNIT NUMBERS
Aerospace Medical Division	62202F
Brooks Air Force Base, Texas 78235	7164-16-13
U. CONTROLLING OFFICE NAME AND ADDRESS USAF School of Aerospace Medicine (VNL) April 1975
Aerospace Medical Division (AFSC)	13 NUMBER OF PAGES
Brooks Air Force Base, Texas 78235	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \
14 MONITORING AGENCY NAME & ADDRESSAL deliterate from	Controlling Office. 15 SECURITY CLASS for this copy of
	UNCLASSIVIED
	154. DECLASSIFICATION DOWNSHAUING
	SCHEOULF
Approved for public release; distribut	1
Approved for public release, distribut	on uniimited.
· •	
	**
17 DISTRIBUTION STATEMENT (of the abstract entered in Blo	ck 20, if different from Report)
18. SUPPLEMENTARY NOTES Reproduced by	
NATIONAL TECHNIC	AL
INFORMATION SERV US Department of Communic Springleds, VA 22151	ICE
	1
Cryogenic sampling, turbine exhaust, as	tily by block number)
exhaust sampling, combustor exhaust sam	polition, politicion analysis,
	Fig. Contacts in a rectified analysis
•	
20 ABSTRACT (Continue on reverse side it necessary and iden Cryogenic sampling was used to sample h	ily by block number
engine combustor under conditions simul	ating idle power of several different
aircraft. Parameters studied were fuel	typeJP4. JP5, and JP8and the combustor
operating pressure15, 33, 50, and 75	psig. Samples were analyzed with a gas
hydrocarbon content of the arbons we	stem which separated 148 compounds; the
hydrocarbon content of the exhaust was pressure and directly proportional to t	he boiling point and density of the fuel
type.	colling point and density of the fuel

DD . FORM 1473 EDITION OF I NOV 65 IS OBSOLETE

UNCLASSIFIED

HYDROCARBON CONSTITUENTS OF T-35 COMBUSTOR EXHAUST

INTRODUCTION

To provide baseline data for assessing the biomedical impact of aircraft operations, the United States Air Force School of Aerospace Medicine (USAFSAM) initiated a cooperative effort with the Air Force Aero Propulsion Laboratory (AFAPL) in late 1972 to conduct sampling and analysis of turbine engine exhaust hydrogarbons from the AFAPL single-combuctor test facility. The approach involved both grab sampling of exhaust, in pressurized containers, and long-term, online exhaust sampling with the USAFSAM multistage cryogenic trapping system. Collected samples were analyzed with a coupled gas chromatograph-mass spectrometer-data (GC-MS-data) system. A preliminary sampling test was conducted in May 1973 to determine feasibility of exhaust cryotrapping and to establish analytical methodology for identifying individual hydrocarbon compounds (3).

This report details results from the second sampling test, conducted in January 1974, to identify and quantitate hydrocarbon emissions from the T-56 combustor as a function of fuel type and combustor operating pressure. The T-56 engine is of the turbine-driven propeller type (turboprop) used on the Air Force C-130 transport aircraft.

EXPERIMENTAL

Test Parameters

The AFAPL combustor rig used for this study (Fig. 1) consisted of a single T-56 series I combustor installed in equipment simulating the airflow characteristics in the actual engine. Compressors and a nonvitiated heating system supplied air at appropriate temperatures, pressures, and flow rates for the experiment.

Rather than perform the entire experiment at combustor inlet conditions corresponding to T-56 engine idle, a number of temperature/ pressure conditions simulating idle operation of a wide range of engine types were tested. Table 1 lists the combustor operating conditions for each test. The lowest temperature condition tested involved 93°C inlet temperature and approximately 15 psig pressure. This corresponds to a number of Air Force engines, such as the J-85, which have overall pressure ratios. The midtemperature setting was at about 166°C, and 33 psig, corresponding to the conditions of many moderate-pressure ratio engines (1). Finally the high-temperature condition of 204°C and 50 psig simulates many newer high-pressure-ratio engines, like the F-101 and F-100. Table 2 lists typical military engines and their respective idle-pressure ratios (4).

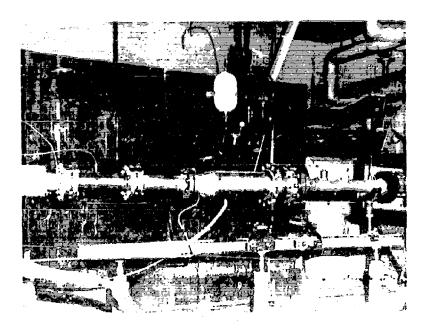


Figure 1. Combustor Test Rig.

The fuels used were JP4, JP5, and JP8. The present Air Force jet fuel, JP4, was tested at all the conditions described above. The current Navy jet fuel, JP5, and JP8, a fuel similar to that used by commercial airlines and occasionally by Air Force aircraft, were tested at an operating pressure of 3.2 atm.

Sampling

The combustor exhaust was continuously sampled from a 5-point sampling rake located approximately $10.2~\rm cm$ (4 in) aft of the combustor liner exit. Hot exhaust gas ($121^{\rm O}{\rm C}$) was delivered, via electrically heated 6.4 mm ($1/4{\rm -in}$) OD stainless steel tubing to online instrumentation for continuous analysis of CO, CO₂, NO_x, and total hydrocarbons (THC), as well as to two 3-stage cryogenic sampling systems. Cryogenic sampling was initiated after combustor operation had stabilized, and was continued until flow stopped because of ice blockage in the CO₂ trap. Nominal sampling time ranged from 60 to 90 minutes. The pressurized grab samples were obtained directly from the sample line at essentially the combustor operating pressure. The pressurized samples were collected in the 10 minutes prior to obtaining the integrated 60-90 minute sample with the cryogenic sampling system and from the same sample line.

TABLE 1. SINGLE-COMBUSTOR RIG OPERATING CONDITIONS

Nominal Power Setting	Rig Pressure psig	Inlet Temp.	Fuel/Air Wt. Basis	<u>Fuel</u>
Preliminary	75	93	.0076	JP4
Moderate PR ^a Simulated idle	33	166	.0070	.JP4
Low PR Simulated idle	15	93	.0073	.1P4
Moderate PR Simulated idle	33	169	.0083	JP5
High PR Simulated idle	50	204	.0079	.1P4
Moderate PR Simulated idle	33	164	. 0071	JP4
Moderate PR Simulated idle	33	166	.0072	JP8
No fuel flow (Background)	33	بت مناس	the gai to any TV	~~~

^aPressure ratio.

TABLE 2. AIR FORCE TURBINE ENGINES

Engine	Aircraft	Idle-pressure ratio
J~85	T-38, F-5A	1.49
J-79	F-104, F-4D, F-4E	2.55
J~57	B-52F-G, F-100, F-101, KC-135	2.60
J-52	C-9	2.80
TF-39	C-5A	3.05
T-56	C-130	3.50
TF-30	F-111	3.56
F-101	B-1	4.0
r-100	F:-15	4.4

The cryogenic sampling system is shown schematically in Figure 2. The sample gas was passed through a flow meter at 500 cc/min (measured at 21.1° C and 760 Torr) into the first trapping cylinder (maintained at 0° C with ice water), through a heated inlet into the second cylinder (maintained at -78° C with pulverized dry ice), through, the final cylinder (maintained at -175° C with liquid nitrogen), and then exhausted to ambient. Two unique features of the cryotrap system are the heated inlet in the -78° C trap to minimize ice formation and a gaseous nitrogen flush in the -175° C trap matrix to prevent oxygen condensation (2).

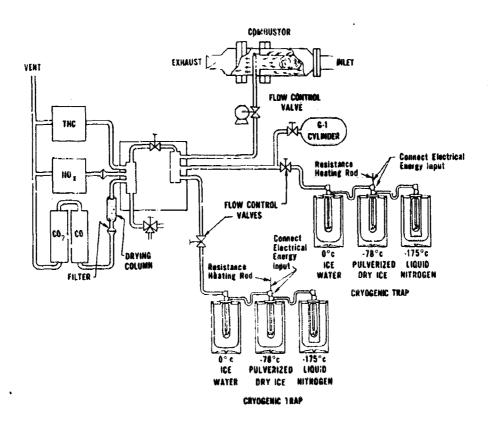


Figure 2. Sampling schematic system.

Analysis

Hydrocarbon analysis of both the pressurized tank and online cryogenic samples was done with a coupled gas chromatograph (Varian nodel 1400) - mass spectrometer (Dupont model 21-491) - data (Dupont 21-094) system (Fig. 3). The chromatographic column packing was Porapak Q (120-150 mesh), in a 3-m-long by 1.6-mm-diameter microbore (0.7 mm) stainless steel tube. This column, with temperature programming, has proven adequate for separating hydrocarbon compounds ranging from ethane to C10 aliphatic and aromatic oxygenates. The chromatographic effluent was split 25% to the chromatographic flame ionization detector (FID), and 75% to the mass spectrometer for sample enrichment via jet separator. Compound quantitation was done by digital integration (Autolab IV) of the chromatograph-FID peak areas. All quantitation of the FID was based on benzene response; i.e., calibration was done with standard gas mixtures of known benzene concentration. Compound identification was done by the MS-data system, based on the data library comparison of fracture patterns (5).



Figure 3. Laboratory analytical system; coupled gas chromatograph-mass spectrometer-data system used for analysis of hydrocarbon compounds in turbine engine exhaust.

The extremely low concentration of hydrocarbon in both cryogenic samples, and particularly in the grab samples, necessitated a laboratory concentration procedure that involved expanding the collected sample through a GC sample loop (2.9 ml) at liquid nitrogen temperature (-196°C). Heating the cryogenic sample bottle to a wall temperature of 150°C made the concentration procedure more effective.

The chromatographic analysis was done by temperature programming in two stages. Before injection, the gas chromatograph was cooled to -100° C. After injection, the air peaks were permitted to clear the column, and the oven temperature was raised to 0° L at approximately 10° C/min. The gas chromatograph was then instrumentally programmed at 10° C/min to a final temperature of 250° C. A terminal isothermal hold was maintained until no additional peaks eluted from the column. The chromatographic carrier gas was helium at a flow rate of 30 cm³/min. The flow rates of air and hydrogen were 350 and 30 cm³/min respectively.

RESULTS AND DISCUSSION

Effects of Fuel Type

Table 3 lists the exhaust hydrocarbon concentrations detected with JP4, JP5, and JP8 fuel, at a constant inlet pressure of 33 psig. Between the three fuels, several differences were noted in both total hydrocarbon content and compound distribution. Compared to JP4, the total concentration of exhaust hydrocarbon was greater with JP5 and JP8, which appeared to show a relationship between exhaust content and fuel density and/or boiling range (JP5 and 8 are higher boiling and density fuels than JP4). The greatest amount of hydrocarbon (5.07 ppm) was obtained with JP5, which has the highest density of the three fuels.

The qualitative differences between the fuels were primarily in the otefins and ketones, with resser shifts in the paraffins, aromatics, ethers, and aldehydes. Overall, the olefins and ketones increased with fuel number, suggesting again a relationship between these compounds and fuel density and/or boiling range. These increases in olefins and ketones were offset somewhat by a decrease in the paraffin and aromatic content although no dramatic change occurred in the concentration of any particular compound. The maximum concentration of both ethers and aldehydes occurred with JP5 fuel, which indicated a trend toward imcomplete combustion with the higher density fuels. The mean molecular weight showed an increasing trend (with fuel number) for the napthenes, aromatics, and aldehydes, and no discernible difference for the other compound classes.

TABLE 3. EXHAUST HYDROCARHON CONCENTRATION WITH VARIOUS FUELS (33 psig combustor pressure)

			FUEL		
COMPOUND (ppm)	JP4 ^a	JP4 ³⁴	JP5	<u>.118</u>	HORE b
PARAFFINS	.83 ^c	.88	.18	.07	<u>u</u>
Methane	.01	.02	.1	.01	
n-Butane				td	
2,2,3-Trimethylbutane	.05				
2,4-Dimethylpentane	.13	10	t		
3-Methylhexane 2,3,4-Trimethylpentane	,02	.19			
3,4-Dimethylhexane	.02	.09			
4-Methylheptane	.02	.09		ŧ.	
3,3-Dimothylhexane				.06	
7,4-Dimethylhexane	.09	.22	.04	• 1013	
3,4-Dimethylheptane	.11		.174		
n-Octane	***	.13			
Unknown		• • •	ι		
n-Nonane	. 65		•		
4-Methyloctane		.15			
2-Methyl-4-ethylhexane	.13	.94			
Unknown	.07		.03		
2-Methylnonane			.01		
Unknown	.04				
n-Decane	. 04	.03			
2-Methyl-5-ethylheptane	.04	.02		t	
3,3,5-Trimethy1heptane				Ł	
Unknown	.02				
Unknown	.03				
Unknown	:				
Avg molecular wt (paraffins)	124.3	107.83	155.92	98.16	
OLEFINS	.76	<u>.76</u>	1.32	<u>1.71</u>	0
Ethylene	.01	.01	.03	.09	
Acetylene	.02	.01	t	.02	
Propylene	.36	.44	.62	.65	
Propyne	.01	.01	t	. 02	
2-Methylpropere	.02	.01		t	
1-Butene	.13	.14	.16	.19	
2- But yne				ŧ	
2.Butene-cis	.01				
1-Pentene			.43	.43	
2-Methyl-1-pentene		.1.			
1-Hexene	- 15				
Unknown	0.0	.02			
3-Heptene-trans	.0.				
2-Methyl-3-Hexene-trans 1-Heptene	t		e e		
2,3-Dimethy1-2-butene			.08	•	
1-Octene			_	.06	
l-Nonene			t	.24 .01	
Unknown	.03			-11	
Avg molecular wt (olefins)	65.93	52.11	59,11	63,2	
D (ATPLANA)	00.75	J = 1 1 1	22111	07.2	

arunlicate combustor conditions background sample (no fuel)

 $^{^{\}text{C}}\textsc{Total}$ by cherical class $^{\text{d}}\textsc{Trace}$, concentration less than 0.001 mam

TABLE 3. (COSTINUED)

Amin 34 (1.91 tunen)					
COMPOUND (ppm)	րկո	JPhila	JP5	3PB	aoanb
DIOLEFINS	<u>.10</u> €	.01	<u>t</u> d	.02	<u>0</u>
Allene	.01	.01	_		
1,3-Butadiene	t		t	.02	
linknown	.09	t	t		
1,2-Pentadiene	.09				
2, 3-Pentadicae		t			
1,3-rentadione-els				t	
Avg molecular wt (diolefins)	59.72	67.03	t si sa		
	39.72	54.07	54.07	54.06	
MAPHEMES	<u>r</u>	.09	.20	.02	<u> 3</u>
Cyclobutane		£.			
Isopropyleyelopropane		.02			
1,2-Dimethylcyclopropane-cis	t	.172			
Cyclohexane	L		10	01	
Methyleyclohexane		0.7	.19	.01	
1-Methyl-2-othylevelopentane-cis		.07			
1,1,2-Trimethylevelohexane			.01		
Avg molecular of (naphthenes)		70.11		.01	
TVE MOTER HAST WE CHAPTERENESS	70.07	79,46	98.14	105.2	
AROMATTCS	.43	. <u>37</u>	.28	.10	$\widetilde{0}$
Unknow :			02		
Ethylbenzene	.15	04	.02		
1,3-Dimethylbenzene	.15	.06	0.0		
1,4-Dimethylbenzene	1.7		.06		
1,2-Dimethylbenzene	.17	.17	.11	.10	
1,2,3-Trimethylbenzene			- 05		
		.04			
n-Propylbenzene m-Ethyltoluene	.08	.04		t	
m-renyitoinene			.03		
1,2,4-Trimethvibenzene		,04			
Unkaewn				t	
1, 3-Diethylbenzene	.03	.02	.01		
Avg molecular wt (aromatics)	106.33	109.79	112.78	116.22	
ALDEHYDES	<u>.62</u>	.81	2.72	1.59	<u>o</u>
Annest Johns					-
.Acctaldehyde	.58	.59	2.07	1.01	
Acrolein			.09	t	
Propionaldehyde	.02	.16	.28	.31	
Crotonaldehyde		.05	.12		
Butvraldehyde	.02	.01	.14	.14	
2-Methylpentanal			.02	.08	
Unknown			r		
n-llexanal	t				
Benzaldehyde			t	. 01	
2,4-Dimethylpentanal				.04	
Avg molecular wt (aldehydes)	t8.59	61.06	77.84	78.95	
			,,,,,,	,	
ALCOHOLS	.06	.14	.03	.33	.09
Methanol					-
Unknown	t	t		t	
Unknown	t				
	.05				٠09
Unknown	t_				
Cyclohexanemethanol	.01				
Isooctylalcohol			.03		
Unknown				.33	
2-PropyTheptanol		.14		t	
Avg molecular wt (alcohols)	107.74	95.09	130.13	116.16	
				• • •	

anaplicate combustor conditions brackground sample (no fuel)

Crotal by chemical class drace, concentration less than 0.001 ppm

TABLE 3. (CONTINUED)

TABLE 3. (Courthoup)		_	:TIST.		
COMPOUND (PAR)	$\frac{1}{(11)^{3}\mathbf{i}^{(1)}}$	1134 P	1104	.अम्ब	acarb
KETONES	7.7.E	32	. 11	.4h	.02
2=Butanone	.02	.69	.19	.35	
Unknown	.02 - 1	.01			
Methylpropylketone 2-Pentanone	. 1		.1.2		
3-Hexanone	.04		02		
3-Methyl-2-pentanone				.61	
5-Methy1-2-hexanone	.10				
4-Methyleyclohexanene				.61	
4-Methyl-3-pentenc-2-one Propylbenzylketone		.17		,03	.02
Unknown		.,,		.91	
3-Heptanone	1				
Unknown	• f) ^r *				
Avg molecular wt (ketones)	100.88	108.59	79.11	111.11	
ETHERS	.95	<u>.08</u>	.13	.09	t J
Furan	.0	.03	, 10	. 119	
2-Methylfuran			•		
3-Methylfuran	1.4	,			
2,3-Fpovybutani Isooctyl vinyl ether :		.91			
Avg molecular wt (ethers)	74.07	94.70	74.07	70.66	
ESTERS	ű	.12	ţ	ņ	11
n-Amyl acetate		.12			
Unknown			t		
Unknown			t.		
Avg molecular wt (esters)		130.18	246.17		
NITROGEN-CONTAINING COMPOUNDS	.05	<u>o</u>	.01	<u>t</u> .	<u>t</u> j
n-Valeronitrile			L		
Nitromechane	.01		.01	İ	
Imidazole	.04 64.56		72.06	n/	
Avg molecular wt (nitregen)	193.311		77.00	61.04	
HALOGEN-CONTAINING COMPOUNDS	.05	.34	.01	.01	$\overline{\mathbf{o}}$
1-Fluorohexane	.04	. 06			
Trichlorogrhylene	. 01	۱۱.	t	.01	
l-Fluoroheptane Unknown		, 604	.01	•	
Unknown		.04			
1-Chloro-3-methylbutanc		.19		t	
Avg molecular wt (halogen)	117. 5	149.97	1.54.76	118.52	
LACTONE	õ	<u>o</u>	<u>.01</u>	.01	<u>5</u>
8,8-Dimethylpropiolactone			.01	.01	
Avg molecular wt (lactone)			100.05	100.05	
TOTAL OF ALL HYDROCARBONS	3.17	3.87	5.27	4.41	0.11

^aDurlicate combustor conditions ^bBackground sample (no f --1)

Total by chemical class direct, concentration less than 0,001 $\ensuremath{\mathsf{ppm}}$

Effects of Operating Pressure

Table 4 lists the exhaust hydrocarbon concentrations as a function of combustor operating pressure for JP4 fuel. As expected, the overwhelming effect of increasing pressure was a large reduction in the exhaust total hydrocarbon content (approximately 6-fold for each factor of 2.2 increase in operating pressure). This was almost certainly a direct result of improved combustion efficiency, which obtained from increased combustion temperature as well as operating pressure.

Although inlet pressure had a significant effect on the quantity of exhaust hydrocarbon, there were only minor changes in exhaust quality. As a percentage of total hydrocarbons, increased operating pressure resulted in increased olefinic content primarily at the expense of paraffins. The maximum percentage of partially oxygenated species occurred in the mid pressure range, from 33 to 50 psig, but was greatly reduced at 75 psig, reflecting the greatest degree of oxidative completion. The mean molecular weight of nearly all classes of exhaust hydrocarbon species tended to increase with operating pressure.

Toxicological Implications

Perhaps the most significant aspect of the data, from the toxicological point of view, was the large number of individual compounds detected in turbine engine exhaust. About 150 compounds were identified in the 8 cryogenically collected exhaust samples, with a mean of 38 compounds per sample. Approximately half the compounds were paraffins, napthenes, olefins and diolefins which, in general, would be less toxic than many of the aromatic and oxygenated species. The overall trend toward higher molecular weight compounds and more oxygenates with increasing fuel density portend a potentially more toxic exhaust with heavier fuels. The finding of B, B-dimethylpropiolactone in the exhaust of JP5 and JP8 was perhaps significant in that this compound is a close analog of B-propiolactone, one of the 13 known carcinogens restricted by the Occupational Safety and Health Administration (6). Howeve-, detailed analysis of the full toxicological impact of turbine engines' exhaust hydrocarbon w'll have to be delayed until data from a broader spectrum of engines and operating conditions are available.

Reproducibility

A major accomplishment of this test series was the excellent demonstration of reproducibility between duplicate cryogenic samples. The two samples in question were taken at an operating pressure of 30 msig with JP4 fuel (Table 3). Total hydrocarbon recovery was 3.17 and 3.87 ppm, respectively. Within hydrocarbon compound classes, the concentration reproducibility was on the average plus or minus 0.05 ppm. The total number of compounds detected in the two samples was 58 and 50, respectively, with over 55% commonality of individual compounds.

TABLE 4. EXHAUST HYDROCARBON COMPOUNDS FOR JP4 FUEL AT VARIOUS INLET PRESSURES

COMBUSTOR INLET PRESSURE (psig)

		COMBOSTOR IN	ILUI I MID	OOKE (PEIE)	
COMPOUND (ppm)	15	33ª_	33ª	50	75
PARAFFINS	11.87 ^b	88.	.83	.23	.16
Methane	.02	.02	.01	.01	.01
1,1-Didevteroethane	.02	•1)2	.01	£C.	*****
n-Butane	t				
n-Pentane	t				
2-Methylpentane	.51				.04
2, J-Dimethylbutane	.,,			.01	•••
Unknown	t				
3-Methylpentane	-				.06
2,2,3-Trimethylbutane			.05		
3-Methylhexane	1.22	.19	.13		
2,3-Dimethylpentane					.01
2,3,4-Trimethylpentane			.02		
3,4-Dimethylhexane	.56	.09	.02		
2,4-Dimcthylhexane	1.70	.22	.09	.02	
3,4-Dimethylheptane			. 11		
2,4,4-Trimethylhexane				.05	
n-Octane	4.91	.13		.03	.01
2,6-Dimethylheptane	.78			.02	
n-Nonane	.84		.05	.03	.01
4-Methyloctane	1.01	.14		5.00	
3-Methyloctane				-t	
2-Methyl-4-ethylhexane		.04	.11	. 02 -	100
3,3-Dimethylhexane	A 15 45			and the second	.01
Unknown		0.00	.07		
2-Methylnonane	.20				
2,2,5-Trimethylhexane				. 02	
Unknown			.04		
n-Decane		.03	.04		
Unknown				.02	
-2-Methyl-5-ethylheptane		.02	.04		01
3,3,5-Trimethylheptane	.12		.02	ŧ	01
Unknown			.02	;	
Unknown Avg molecular wt (paraffins)	98.43	107.83	124.3	113.65	94.56
OLEFINS	3.80	.76	,76	.17	.28
			-		
Ethylene	.24	.01	.01 .02	.01	,01 ,05
Acetylene	.28 1.35	.01 .44	.36	.11	.13
Propylena	.05	.01	.01	.11	.13
Propyne	.09	.01	.02		.01
2-Methylpropene	.60	.14	.13	.05	.08
l-Butene 2-Butyne	t	•14	.13	.00	.00
2-Butene-cis	L		.01		t
2-Pentene-cis	.64		• • • •		-
2-Methyl-1-pentene	•	.12			
1-Hexene			.15		
4-Methyl-1-Hexene	.55		• • • •		
Unknown		.02			
3-Heptene-trans			.02		
2-Methyl-3-Hexene-trans			t		
Dipentene			-	t	
Unknown			.03		
Avg molecular wt (clefins)	52.30	52.11	65.93	65.62	44.08

^aDuplicate combustor conditions ^bTotal concentration by chemical class ^cTrace, concentration less than 0.001 ppm

TABLE 4. (CONTINUED)

		COMMETOR	INLET PRESS	SURE (psig)	
COMPOUND (ppm)	_15	33 ^{tt}	_33 ^{&}	50	75
DIOLEFINS	<u>.05</u> b	.01	.10	.02	ود
Allene	.05	.01	.01	***************************************	_
2,3-Butadiene	t	t	t		
Unknown			.09		
1,2-Pentadiene Avg molecular wt (diolefins)		t		.02	ŧ
	54.07	54.07	59.72	68.06	68.06
NAPHTHENES	.53	.09	<u>t</u>	.05	.01
Cyclobutane					
isopropylcyclopropane		t •02			
1,2-Dimethylcyclopropane-cis		102	t		
Unknown			· ·	t	
Methylcyclohexane	.53	.07		. 05	.01
Avg molecular wt (naphthenes)	98.19	79.46	70.07	84.14	98.19
AROMATICS	1 42	2.7			,,,,,
	1.43	<u>.37</u>	<u>.43</u>	.04	<u>.08</u>
Ethylbenzene		.06	1.5		
1,3-Dimethylbenzene		100			.01
1,4-Dimethylbenzene	.81	.17	.17	.02	.01
1,2,3-Trimethylbenzene n-Propylbenzene	.53	. 04			
p-Ethyltoluene		. 04	. 08		
1, 1,4-Trimcthylbenzene				.02	
Unknown	.09	.04			,
1-Methy1-2-Ethylbenzene	.09				
1,3-Diethylbenzene		.02	.03		. 06
Avg molecular wt (aromatics)	116.2	109.79	106.33	99.4	.01 . 106.34
ALDEHYDES					
	3.55	.81	.62	<u>. 17</u>	.04
Acetaldehyde	2.3	.59	.58		
Acrolein	t	133	+36	.16 t	.01
Propionaldehyde	.61	.16	.02	t	t
Crotonaldehyde	.23	.05	•./-	.01	E
Butyraldehyde 2-Methylpentanal	.36	.01	.02	t	t
n-Pentanal	.05				
Unkney				t	
n-Hexanal	t				
n-reptanal			t		
BenzIdehyde				t	t
Avg molecular wt (aldehydes)	73.53	61.06	68,59	t 75.84	.03 77.43
ALCOHOLS	.43	.14	04	•	
		+14	.06	<u>o</u>	<u>t</u>
Methano!	t	t	t		
Unknown Unknown			t		
Unknown Unknown			.05		
Cyclohexanemethanol			t		
2-Propylheptano1			.01		
2-Butyl-1-cotanol	.43	.14			t
Avg molecular wt (alcohols)	109.11	95.09	107,74		100 11
	107111	33.03	107.74		158.16

ADuplicate combustor conditions LTotal concentration by chemical class Carsec, concentration less than 0.001 ppm

TABLE 4. (CONTINUED)

	9	COMBUSTOR	INLET PRESSU	RE (paig)	
COMPOUND (ppm)	15	338	33ª	50	75
KETONES	.62 b	.27	.22	<u>t</u> c	<u>t</u>
2-Butanone Unknown Methylpropylketone 3-Hexanone	.62	.09 .01	.02 .02 t	t	t
Unknown 5-Methyl-2-hexanone Propylbenzylkecone		.17	.10		t
Unknown 3-Heptanone Unknown Avg molecular wt (ketones)	65.07	108.59	t .04	cc 0 *	t
AVE MOISCUIZE WE (RECORES)	03.07		100.88	65.07	76.09
ETHERS	.15	.08	.05	.02	<u>t</u>
Puran 2-Methylfuran	.14 .01	.03 t	.02	.02	
3-Methylfuran 2,3-Epozybutane		.04	.03		t
Lisooctyl vinyl ether Avg molecular wt (ethers)	75.09	.01 94.59	74.08	68.07	72.05
ESTERS	<u>0</u> -	.12	<u>o</u>	<u>o</u>	.02
n-Amyl-acetateHeptyl formateUnknown		.12			.01 .01 t
			7 =		
NITROGEN-CONTAINING COMPOUNDS	· · · <u>o</u> · · · · ·	<u>o</u>	- <u>.05</u>	<u>t</u>	<u>o</u>
Nitromethane Unknown Imidazole		•	.01	t	
Avg molecular wt (nitrogen)			64.56	99.06	
HALOGEN-CONTAINING COMPOUNDS	.62	.34	.05	.03	<u>t</u>
Unknown Chloroethylene	t	9.5		t t	t
l-Fluorohexane Trichloroethylene Unknown	t	.06 .01	.04 .01	t .03	t
Amy1-2,2-dichloropropionate Unknown Unknown 1-Chloro-3-methylbutane	.62	.04 .04 .19		. 33	
Avg molecular wt (halogen)	135.31	149.97	117.75	90.63	96.95
LACTONE	<u>o</u>	<u>o</u>	<u>o</u>	<u>.01</u>	0
2,2-Dimethylpropiolactone Avg molecular wt (lactone)				.01 100.05	
TOTAL HYDROCARBON CONTENT	23.05	3.87	3.17	0.74	0.59

aDuplicate combustor conditions
bTotal concentration by chemical class
cTrace, concentration less than 0.001 ppm

SUMMARY AND CONCLUSIONS

Cryogenic sampling was used to sample hydrocarbon exhaust from a T-56 turbine engine combustor under conditions simulating the idle power setting of several different Air Force inventory aircraft. The parameters studied were fuel type (JP4, JP5, and JP8) and combustor operating pressure (15, 33, 50, and 75 psig). The principal conclusions from the study were:

- 1. Cryogenic sampling was an effective and reproducible technique for sampling gaseous hydrocarbon exhausts from turbine engines.
- 2. The hydrocarbon content of combustor exhaust was inversely related to operating pressure.
- 3. The hydrocarbon content was directly related to fuel density and/or boiling point.
- 4. About 150 compounds were identified; of these approximately half were aromatic and oxygenated species.
- 5. A trend toward higher molecular weight compounds and more oxygenates with increasing fuel density suggested a more toxic exhaust with heavier fuels.
- 6. Duplicate-samples demonstrated excellent reproducibility, with a concentration variation on the average of 0.05 ppm.

REFERENCES

- 1. Blazowski, W. S., and R. E. Henderson. Aircraft exhaust pollution and its effect on the U.S. Air Force. Technical Report AFAPL-TR-74-64, Aug 1974.
- 2. Conkle, J. P., J. W. Register, and G. L. Worth. Multi-stage cryogenic trapping system. Aerosp Med 36:869-874 (1965).
- 3. Conkle, J. P., W. W. Lackey, and R. L. Miller. Cryogenic sampling of turbine engine exhaust. SAM-TR-74-54, Nov 1974.
- Dobbs, M. B., et al. Selected research and development projects in environmental quality. Technical Note AFWL-TN-74-005, Mar 1974.
- 5. Hertz, H. S., R. A. Hites, and K. Biemann. Identification of mass spectra by computer. Searching a file of known spectra. Anal Chem 43:681-691 (1971).
- OSHA, Emergency temporary standard on certain carcinogens. Federal Register 38:10929, May 3, 1973.